

Innovative Solutions Through Test and Analysis-Driven Design

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#### Prepared for:

What's New in Femap 11.3 Webinar

Date:

April 12, 2016

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in ata-engineering

@ATAEngineering

# Outline

- 1. Introduction
- 2. Visualization Example
- 3. Contact Example
- 4. Mesh Refinement Example
- 5. Postprocessing Example
- 6. Solver Support Updates
- 7. Question and Answer Session



## Introduction

v11.3 release scheduled to happen by the end of April

- ➤Look for the email
- ➢ Download on Siemens GTAC website

## ➤Femap Direction

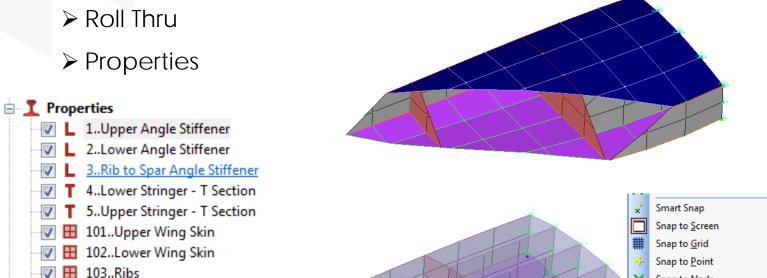
- Dedicated FEA environment
- Detailed functionality to accurately model real-world problems
- Customer driven features and functionality



### Visualization Example

Wingpost Part 1

- Draw/Erase Toolbar
- ➢ Pick Front
- ➢ Roll Thru
- ➢ Properties





104...Spar Webs

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Snap to Node

Pick Normal

Pick Query

Pick <u>F</u>ront

Select Rotate View...

Pick Any Inside Pick All Inside

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## Contact Example

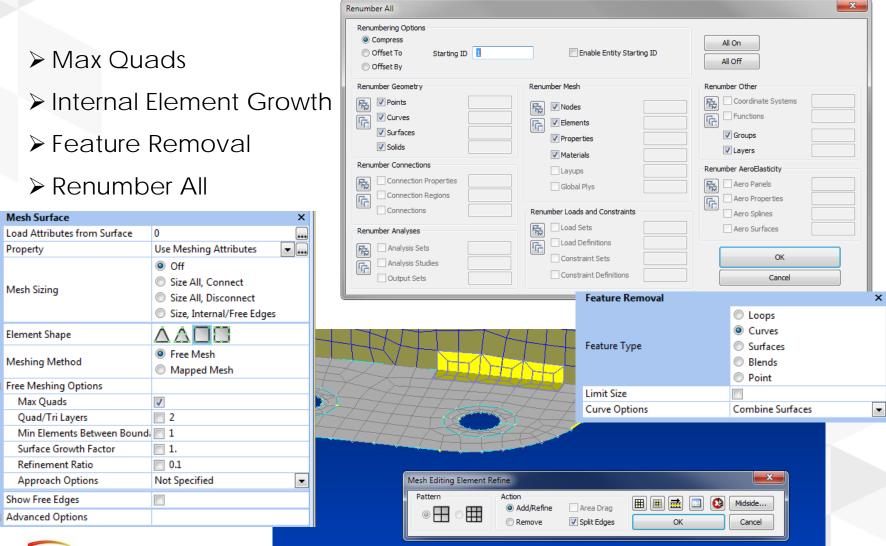
#### Connection Manager

#### Element Face Picking

	n Editor							
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D	Status T	itle	Property	Master(Target)	Slave(Source)	X Center	Y Center	Z Center
	🗹 C	onnect_1	1Contact_1	1Lower	2Upper	9.	2.	0.255
		1111						
							23	2
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			Selection	n Info ement 5832	Face 4		Faces Selected	
		Method	Faces Ele		Face 4	From 608 Elements	aces Selected	
		Method O Adjacent	Faces Ele I Faces Toler	ement 5832	Face 4		Faces Selected	
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		Method	Faces Ele Taces Toler face rdinates	ement 5832	Face 4		Faces Selected	



### Mesh Refinement Example

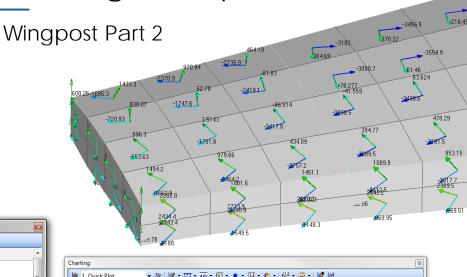


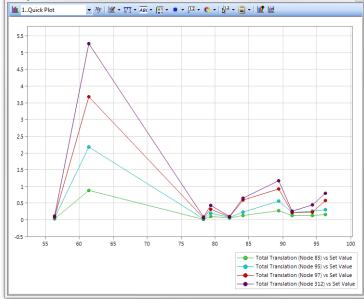


### Postprocessing Example

- Contour Arrows
- Output Set Titles
- Charting Analysis Studies

0.00		
	김 글 • 🦨 🚱 🗄 입니	
Element	7026Plate Top MajorPrn Str	ess 7027Plate Top MinorPrn Stress 🕢
100	-1247.733	-5708.224
103	-1197.979	-5455.101
97	-932.6013	-5269.575
106	-833.1339	-4868.402
101	-309.2333	-4764.263
104	-328.9654	-4586.746
88	-545.8931	-4083.238
102	-505.8874	-4051.597
98	117.9048	-3949.435
105	-509.0112	-3912.452
91	-546.6316	-3902.092
85	-216.4921	-3856.371
99	-219.4901	-3836.535
89	61.45992	-3554.92
94	-194.0677	-3552.684
108	-201.3204	-3551.229
86	378.2191	-3456.518
92	63.62388	-3438.628
107	124.7053	-3432.037
87	304.6888	-3182.041
95	478.295	-3181.51
90	-78.27747	-3090.708
96	394.7728	-3089.455
273	-99.45749	-3026.931
93	-41 55893	-2998 457

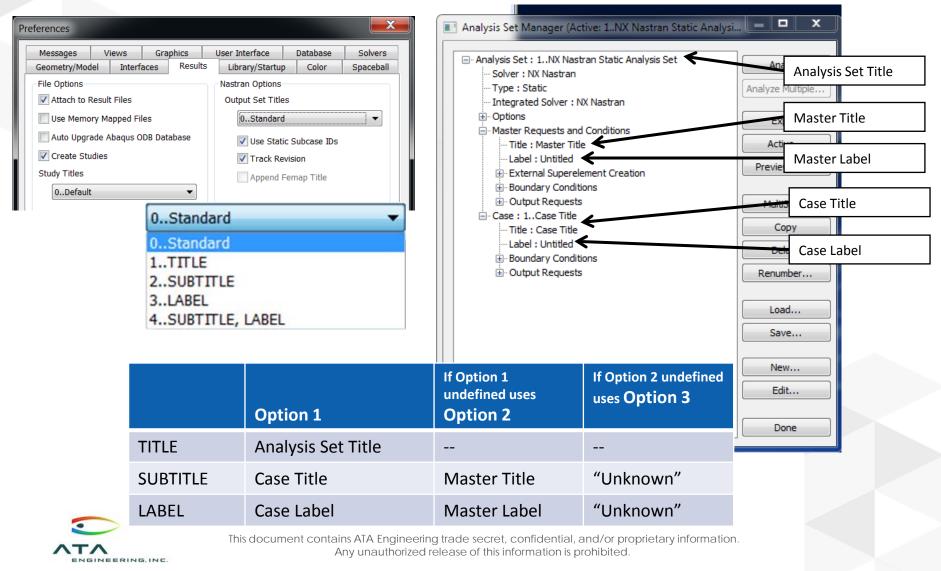






### Postprocessing

#### **Results Titles**



Abaqus ODB Files

- First time support of the ODB file
- Attach to Abaqus results
  - Internalize selected results
- Supports prior version ODB
- Works like currently supported results formats including Nastran OP2 and XDB, CSV, and Femap FNO
- ➤ Fully compatible with all postprocessing functionality

	0 Attached File(s)		Attack Options
ID	Filename	File Path	Attach Options File Format NX Nastran MSC/MD Nastran NEI Nastran/Femap FN
			Abaqus     Memory Mapped File



#### Non-zero Constraints

#### > Non-zero Constraints aka Enforced Displacement

#### ➢ SPC in constraint set

Constraint Set 1	Cantilever		
Title		Coord Sys 0Basic Rectangular	•
Color 120 Palette	e Layer 1		
DOF		X Symmetry X AntiSym NonZero Cor	nstraint >>
TX TY TZ	Fixed Pinned	Y Symmetry Y AntiSym	
RX RY RZ	Free No Rotation		`

Create No	nZero Nodal Constraints	X
Constrain	t Set 1 Cantilever	
Title	Non-zero SPC	Coord Sys 0Basic Rectangular
Color	120 Palette Layer 1	
DOF TX TY TZ	0.01         RX         0.           0.05         RY         0.           0.         RZ         0.	Fixed     Pinned     X Symmetry     X AntiSym       Free     No Rotation     Y Symmetry     Y AntiSym       Z Symmetry     Z AntiSym
	DOF Only >>	OK Cancel

\$	Femap	with	NX	Nastran	Constrair	nt Set	2	:	Enforced	Displacement	
a	ъа			۰ ۱ <i>۵</i> ٬	<b>`</b> 1		71				

SPC	2	103	1	.01
SPC	2	103	2	.05



#### NX Nastran Solution Control

- Additional command line access
- ➢ Default from file → preferences
- Additional GPU control
  - More options as they become available

NASTRAN Executive and Solution Options	X									
Solver Direct Output To Base Filename for Analyze (Blank to Match Model) Additional Command Line	SMEM=10GB parallel=4									
Executive Control Problem ID Solution Override	MSC/MD Nastran Version Ver 2001 Previous Versions									
Max Time (in minutes) 10000.	Solution Options Iterative Solver 00ff									
System Cells           System Cells           Extended Error Messages           Extended Solution Status Monitoring	✓ Number of Processors       4         Solver Memory (Mb 0=Auto)       0         ✓ GPU Computing       1DCMP									
<u>E</u> rom	Restart Control          Save Databases for Restart         Restart Previous Analysis									
Version Starting Su Manual Control	Start Text (Off) End / DMAP Text (Off)									
Prev Ne <u>x</u> t Scratch Files.	<u>O</u> K Cancel									



#### CBUSH/PBUSH in ANSYS and ABAQUS

- No matching element in ANSYS and Abaqus
- Helps when transferring models from Nastran
- Calculates stiffness and damping entries according to the CBUSH/PBUSH input
- Uses a general stiffness matrix
  - ➢ MATRIX27 in ANSYS
  - \*MATRIX INPUT in Abaqus
- Creates spring (K) and/or damper (B) elements
- Matrix is calculated once at export, so nonlinearity and frequency dependencies are ignored

Define P		_	RING/DAMPE		1	Materia		۲operty Type
	© <u>C</u> BUSH		Other(N	IASTRAN CRO	D/CVISC)			
	ty Values	Property	Torsiona /alues	l Stiff	ness 0.		Dam	iping 0.
<u>D</u> OF 1 2 3 4 5 6	Stiffne 100. 200. 300. 400. 500. 600.		Damping 10. 20. 30. 40. 50. 60.	Structural D 1. 2. 3. 4. 5. 6.	Damping	Ori <u>e</u> nta	/Damp Loc ation CSys rain Recover Stress Coe 0. 0.	
Nonlin	near/Fred	<sup>sp</sup>		<u>S</u> ave.		<u>ny</u>	QK	Cancel



#### New Element Types

### ➤ General Matrix

NGINEERING.INC.

- ➤ 12x12 matrix (K,B,or M)
- Added primarily for CBUSH/PBUSH support in ANSYS and Abaqus

	ystem for Matri <u>x</u>	0Basic Re	ectangular	•	Matrix <u>T</u> ype	0Stiffness	•			6 <u>x</u> 6 M	Aatrix
- Matrix <u>V</u> alues 1	2	3	4	5		0Stiffness 1Damping 2Mass		9	10	11	12
100.	0.	0.	0.	0.	0.934779	-100.	0.	0.	0.	0.	0.934779
	200.	0.	0.	0.	-49.96504	0.	-200.	0.	0.	0.	-49.96504
		300.	-2.804336	74.94756	0.	0.	0.	-300.	-2.804336	74.94756	0.
			400.0262	-0.700594	0.	0.	0.	2.804336	-399.9738	-0.700594	0.
				518.7238	0.	0.	0.	-74.94756	-0.700594	-481.2762	0.
					612.4913	-0.934779	49.96504	0.	0.	0.	-587.5087
						100.	0.	0.	0.	0.	-0.934779
		symmetric					200.	0.	0.	0.	49.96504
								300.	2.804336	-74.94756	0.
									400.0262	-0.700594	0.
										518.7238	0.
											612.4913

#### New Element Types

### ➢ Spring to Ground

- CBUSH or DOF Spring elements
- For zero length or grounded elements, CID must be specified

Parabo	lic Elements
Line Elements	Plane Elements
Rod	Shear Panel
🔘 Tube	Membrane
Curved Tube	Bending Only
🔘 Bar	Plate
🔘 Beam	C Laminate
🔘 Link	Plane Strain
Curved Beam	Axisymmetric Shell
Spring/Damper	Plot Only
DOF Spring	Volume Elements
Gap	
Plot Only	Axisymmetric
	Solid
	Solid Laminate
Other Elements	
Mass	Rigid
Mass Matrix	General Matrix
Spring/Damper to Group of the second seco	und 🔘 Slide Line
DOF Spring to Ground	Weld/Fastener
Element Material Orienta	tion OK
Formulation	Cancel

		IG/DAMP->GRC	OUND Element T	···						
ID 9		BUSH to Ground	Material ▼							
	<u>C</u> olor 1	10 Palette	Elem/Property Type							
NASTR	NASTRAN BUSH Property Values									
DOF	Stiffness	Damping	Str <u>u</u> ctural							
1	15000000.	0.	0.	Spring/Damp Loc 0.						
2	1000000.	0.	0.	Orientation CSys 0Basic Rectar -						
3	1000000.	0.	0.	Stress/Strain Recovery						
4	0.	0.	0.	Stress Coef Strain Coef						
5	0.	0.	0.	Trans 0. 0.						
6	0.	0.	0.	Rot 0. 0.						
Nonlin	near/Freq Resp	Loa <u>d</u>	<u>S</u> ave	lopy <u>Ω</u> K Cancel						

< 1	><	2	><	3	><	4	><	5	><	6	><	7	><	8	><	9	><	10	>
CBUSH	13199				9	70	)89										0		
CBUSH		13200			9	88	310										0		
CBUSH		13201			9	105	531										0		
CBUSH		13202			9	122	12252							0					
CBUSH		13203			9	139	973										0		
CBUSH		13204			9	24	183										0		
CBUSH		13205				3647								0					
CBUSH	13207				10	53	5368							0					



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